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Challenges in Assessing Orbital Debris Impact Risk: IDA Support from the Columbia Failure Investigation to Today

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Joel E. Williamsen

September 2020

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IDA Document NS D-14305

Log: H 2020-000299/2

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About This Publication

This work was conducted by the Institute for Defense Analyses (IDA) under contract HQ0034-19-D-0001, Task C1330, "Space Issues Working Group," for the Office of the Director, Operational Test and Evaluation. The views, opinions, and findings should not be construed as representing the official position of either the Department of Defense or the sponsoring organization.

Acknowledgments

The IDA Technical Review Committee was chaired by Mr. Robert R. Soule and consisted of Dr. Daniel L. Pechkis from the Operational Evaluation Division, and Dr. Asha Balakrishnan from the Science and Technology Policy Institute.

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Executive Summary

This presentation will be part of the planned “IDA Forum on Orbital Debris Risks and Challenges,” to be held in September 2020 for IDA attendees and selected government officials from the Executive Branch. It summarizes the orbital debris risk assessment process as developed and employed by NASA. It also describes IDA research and support in various orbital debris risk assessment tasks performed for NASA and other organizations, from the Columbia disaster in 2003 to the present.

Challenges in Assessing Orbital Debris Impact Risk: IDA Support from the Columbia Failure Investigation to Today

Dr. Joel Williamsen

Operational Evaluation Division

Institute for Defense Analyses

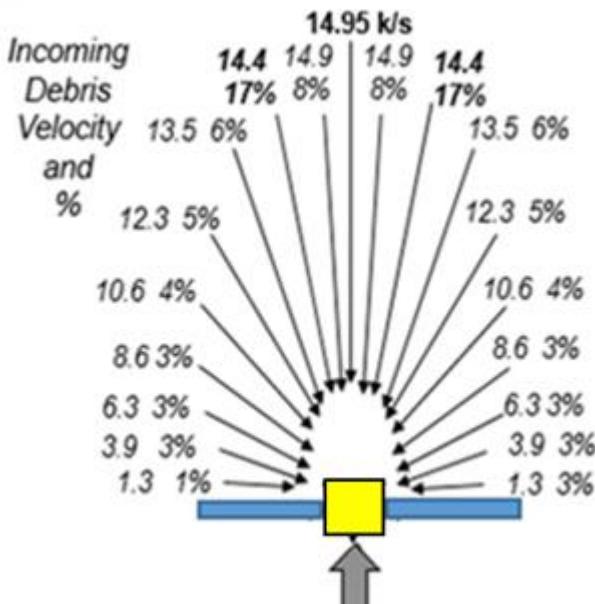
July 20, 2020

Untracked Orbital Debris Poses a Great Risk to Active Satellites

Without warning, small untracked orbital debris from 1mm to 5mm can kill satellites in low earth orbit (LEO), where rapid satellite population growth is taking place

Untracked debris is a much higher risk than tracked debris to active satellites

Debris Velocities at 800 km Orbit



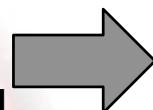
Most debris approaches “front” of spacecraft – and is the highest velocity

High impact velocities drive potential spacecraft damage

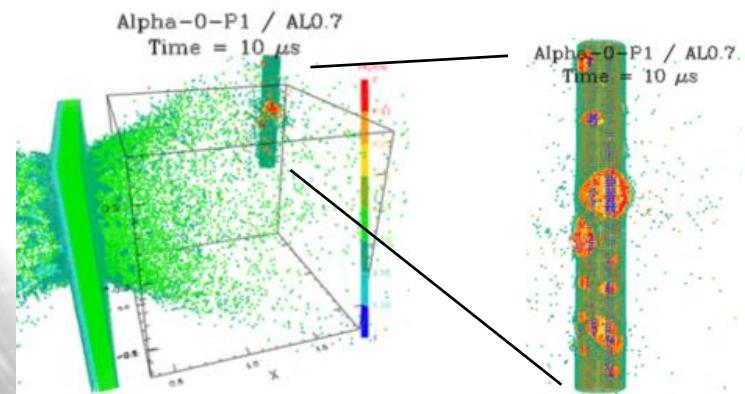
Orbital Debris

Impact Energies at 14 km/s

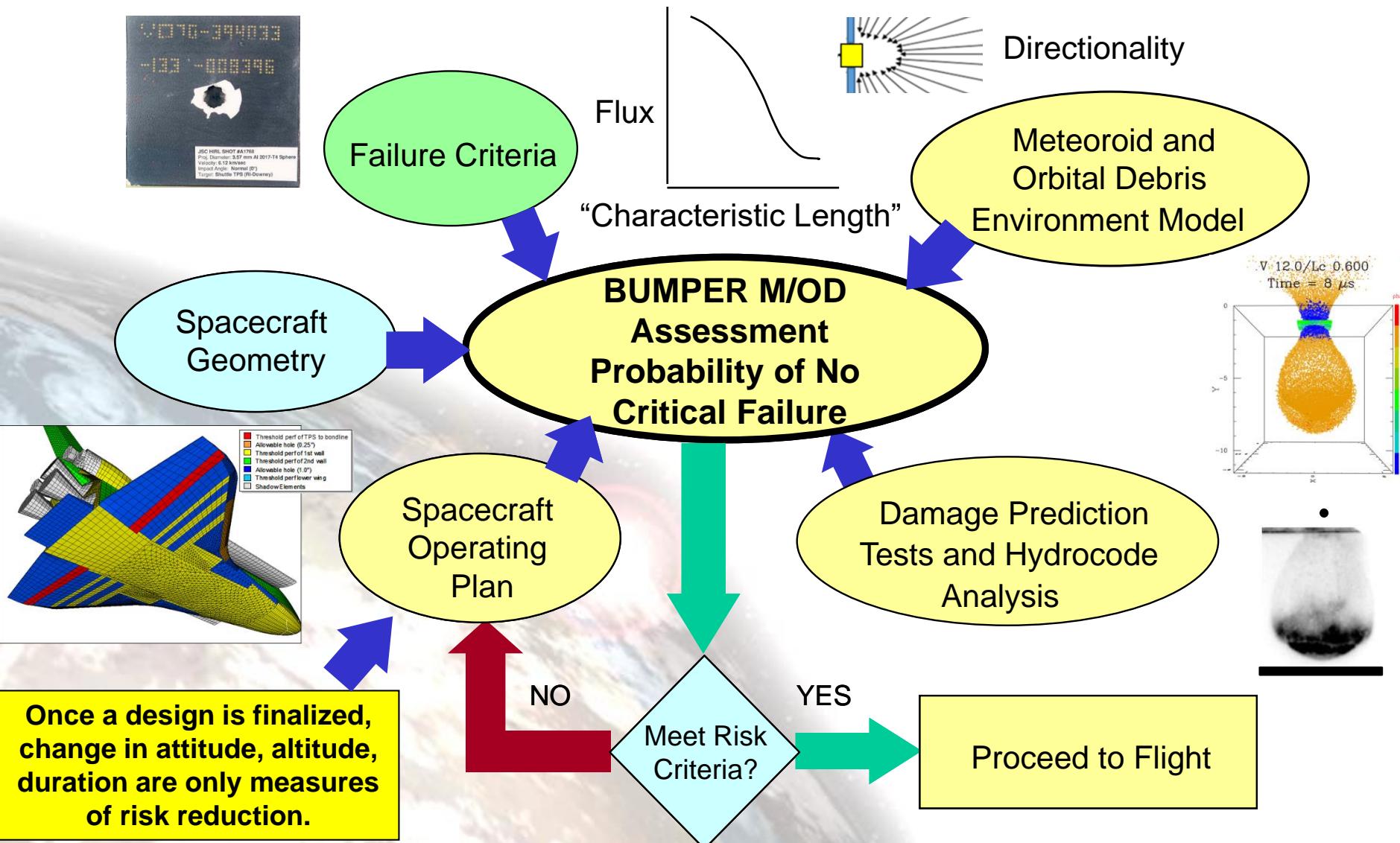
Alum Sphere	1mm (.0015g)	3mm (.039 g)	1 cm (1.5 g)
Energy (j)	192	4992	192,000
TNT (g)	0.45	1.19	45.9
Other Equiv.	.22 bullet	.30-06 bullet	Mark 2 grenade



IDA high-speed impact simulation against NISAR spacecraft wiring harness-- 0.7 mm AL particle impacting multi-layer insulation over Kapton covered wires



NASA's Critical M/OD (Meteoroid/Orbital Debris) Risk Assessment Process



2003-2004 Space Shuttle Columbia Investigation

2007-2008 Orion Spacecraft M/OD Design Evaluation

2009-2010 Deep Impact “EPOXI” Mission Support

2011-2012 International Space Station M/OD Investigation

2014 Briefings on M/OD Protection to National Science Foundation

2015-2016 Joint Polar Satellite System Design Evaluation and Review

2016-2017 Evaluation of Orbital Debris Risk Predictions With Available On-Orbit Assets

2018-2019 NASA Support for M/OD Effects on NISAR Spacecraft Wire Harness Design

2019-2020 NASA ORDEM 3.1 Orbital Debris Environment Review

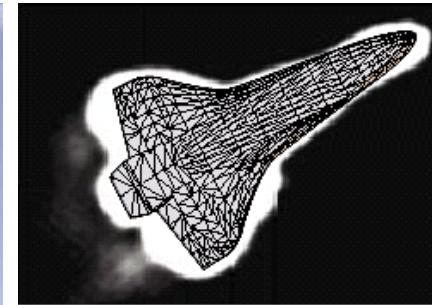
IDA supported many of these studies through the NASA Engineering and Safety Center (NESC), which brings experts from within and outside NASA to provide ongoing independent safety assessments.

The Loss of Columbia Orbital Debris Strike Was a Primary Suspect

On February 1, 2003, Space Shuttle *Columbia* was destroyed in a disaster that claimed the lives of all seven of its crew.

The Columbia Accident Investigation Board was established within two hours of the loss of signal from the returning spacecraft.

One of the potential causes of the loss was a **meteoroid/orbital debris (M/OD) strike** on Columbia's thermal protection system (TPS) while in orbit, causing failure of the orbiter during re-entry.



IDA Supported the CAIB and NASA “Independent Assessment”

**Columbia Accident
Investigation Board
(CAIB)**
Chair: Adm. Gehman

Aim: To conduct “not only an investigation of what happened to *Columbia*, but also – to determine the conditions that allowed the accident to occur – **a safety evaluation of the entire Space Shuttle Program.**”

Examined:

- Launch Debris from ET Foam
- Meteoroid/Orbital Debris Strike
- Willful Damage/Security
- SRB Bolt Catchers
- Kapton Wiring (Arcing)
- Hypergolic Fuel Spill
- Space Weather
- “Rough” Left Wing
- Training and On-orbit Performance
- Payload Malfunction
- Foreign Object Damage

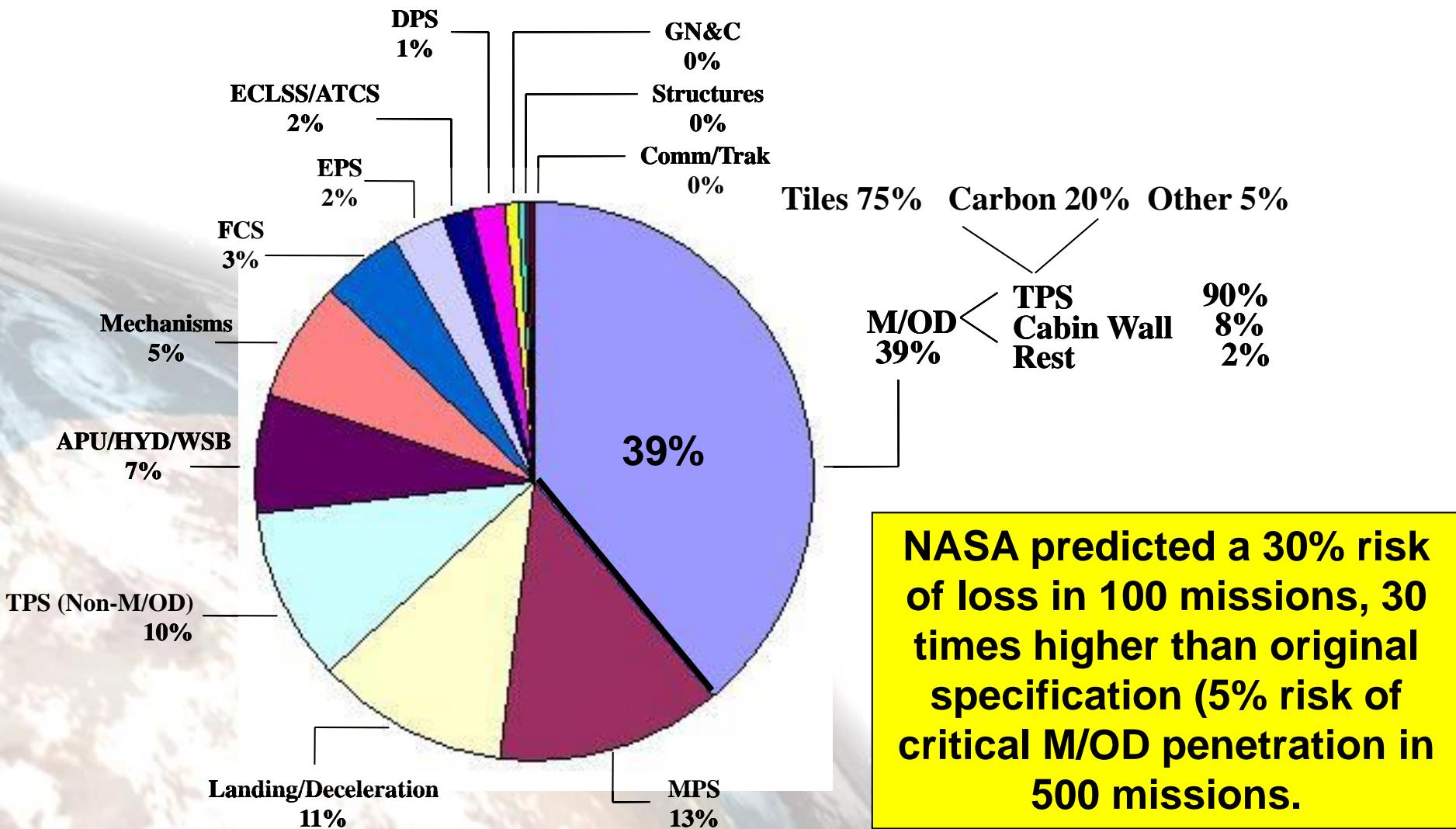
NASA Support (IDA's Direct Client)
JSC Office of Safety and Mission Assurance
Independent Assessment – Hugh Baker

- On-orbit anomalies JSC Mission Ops
- “Day 2” RADAR Object NORAD
- **M/OD Risk Assessment** IDA
 - **What is the risk of an M/OD penetration?**
 - **How does NASA compute the risk for each flight? How does it reduce risk?**
 - **How can this process be improved?**

Report August 2003

M/OD was a Major Candidate for Causing Columbia Loss Finding: Risk Far Exceeded Original Specification

"A Priori" Estimate of M/OD Risk vs. Other Orbiter Risks



CAIB Study Found Inaccuracies in Debris Models and their Prediction Uncertainties

- Meteoroid Environment was Wrong (*fixed, increased risk*)
- NASA Needs a Tool to Quantify Its Prediction Uncertainties (*added, not normally used*)
- OD Environment and Penetration Model Needs Improvement (*still does*). Model approximates debris as spheres, which over predicts their mass and risk.

An IDA theme over many studies:
Equating “characteristic length”
with a spherically shaped particle
adds unneeded spacecraft shielding.



Typical debris from a satellite breakup (DoD SOCIT Tests, 1993).

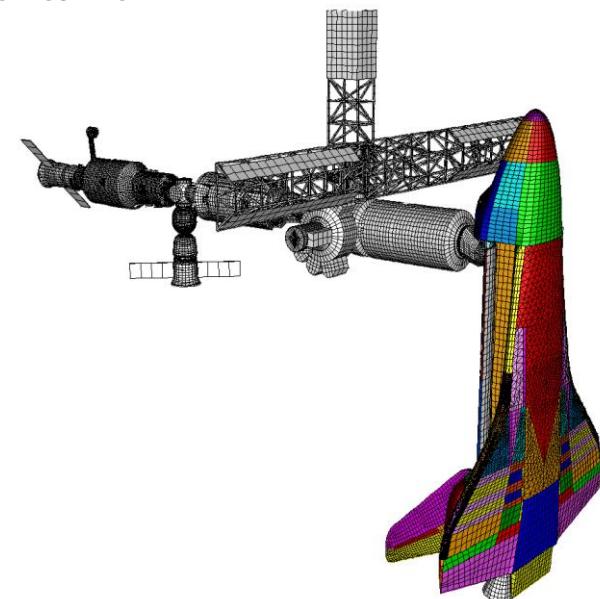
Where are the spheres?

The good news...hypervelocity impact modelling has improved in speed and accuracy. This allows formation of OD models that include shape prediction.

CAIB Study Led to Some Reform, but More is Needed

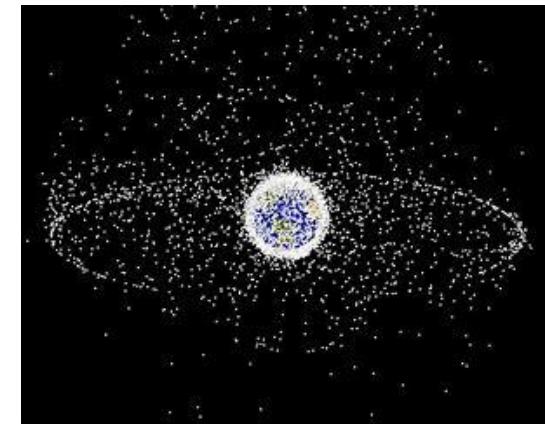
- Independent M/OD safety assessment...isn't (happening).
Added NASA Engineering and Safety Center—outside oversight (like DOT&E).
- The shuttle and station are linked...but not in M/OD analysis.
Started linking them immediately
- XYO (Examine Your Organization)! Threat characterization and risk assessment shouldn't mix
- This wasn't the first time they'd heard this...Recommendations from the 1997 National Research Council Committee on Space Shuttle Meteoroid/Orbital Debris Risk Assessment are still valid. Notably: “NASA should reconsider conducting on-orbit surveys of the orbiter exterior to detect impact damage and repair it if necessary.”

DOCKED CONFIGURATION



Does NASA Overestimate Orbital Debris Risk? Recent IDA-Supported NASA Study Results

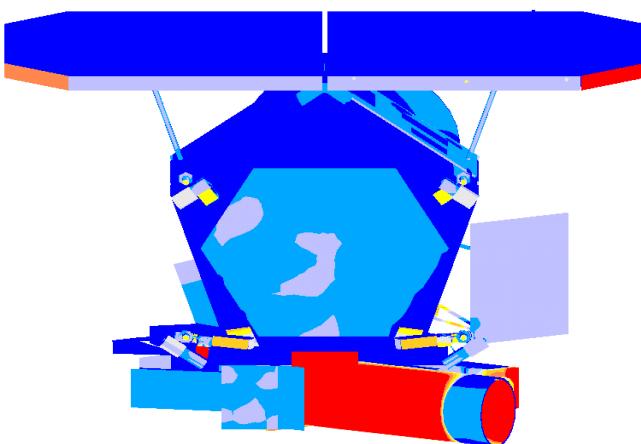
- Considering over 100 LEO satellites over 20 years, **NASA over predicted the number of satellite failures by a factor of ~5** (8 to 11 failures predicted vs. two actual).*
- **NASA and Boeing calculated the number of expected radiator leaks aboard the ISS as between 1.6 and 3.1 perforations, but none occurred** (driven by ~1.5 mm orbital debris particles penetrating vulnerable tubing).
- A well-known satellite constellation reported **only seven movements of > 3 m dMSA**. IDA predicted that **> 10 times more movements of these magnitudes should have occurred if NASA's OD model is accurate**.
- The mismatch in expected versus actual failures could be due to either:
 - Conservative orbital debris environment models (use of spheres vs. flakes, etc.)
 - Conservative penetration model assumptions
 - Anomaly underreporting (not likely based on narrow failure modes focused on)
 - Or, a combination of these factors.



* Squire, M., et al. "Evaluation of Micrometeoroid and Orbital Debris (MMOD) Risk Predictions with Available On-Orbit Assets" (NESC-RP-14-01000). Hampton, VA: NASA Engineering and Safety Center, September 1, 2017.

IDA Has Contributed to Numerous Other Spacecraft Risk Assessment Projects

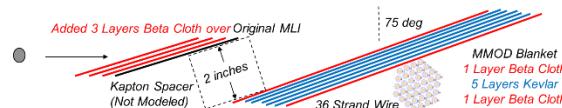
Deep Impact Flyby Spacecraft Expected Element Penetrations: Comets Temple I and Hartley



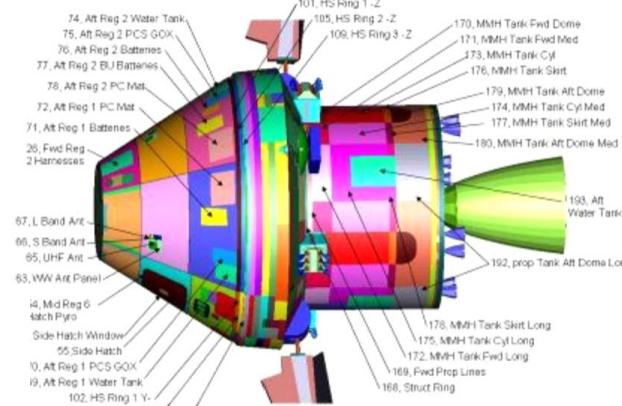
Damage Prediction Models



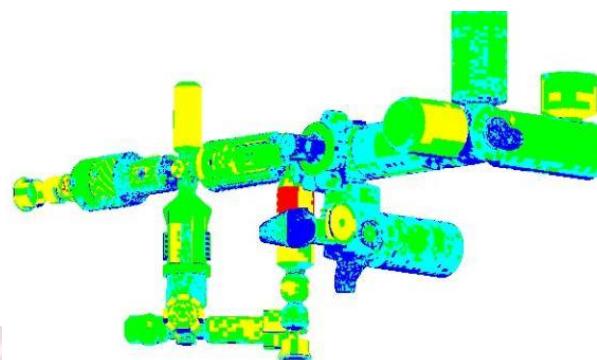
Joint Polar Satellite Shielding



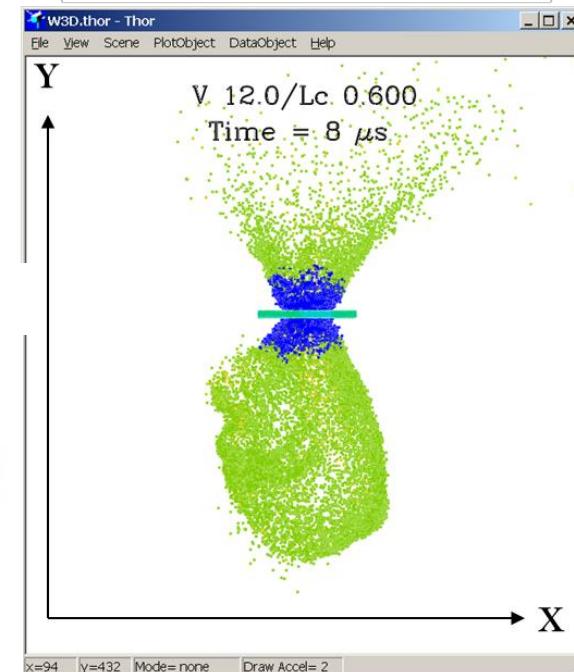
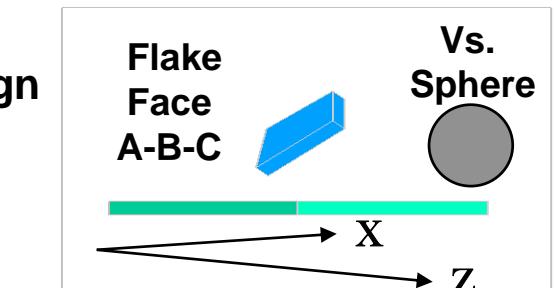
Orion Thermal Protection Design



Space Station Protection



Effect of a “Flake” OD Shape Model on Space Station Risk: Cuts Risk in Half



IDA Continues to Support Spacecraft Risk Assessment

- Continued assistance to NASA Engineering and Safety Center
- Assistance to Commerce Department and NASA
 - Establishing orbital debris data repository
 - Trusted agent for protecting data
 - Potential risk reductions possible
- Support to Office of Science and Technology Policy
 - Identify R&D needs
 - Point to potential FCC risk assessment process improvements
- Assistance to DOT&E and larger DoD – military satellite risk assessment support
 - Meteoroid/orbital debris risks
 - Anti-satellite risk assessment

REPORT DOCUMENTATION PAGE

*Form Approved
OMB No. 0704-0188*

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. **PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.**

1. REPORT DATE (DD-MM-YYYY) 09-2020	2. REPORT TYPE IDA Publication	3. DATES COVERED (From - To)		
4. TITLE AND SUBTITLE Challenges in Assessing Orbital Debris Impact Risk: IDA Support from the Columbia Failure Investigation to Today		5a. CONTRACT NUMBER Separate Contract		
		5b. GRANT NUMBER		
		5c. PROGRAM ELEMENT NUMBER		
6. AUTHOR(S) Joel E. Williamsen (OED)		5d. PROJECT NUMBER C1330		
		5e. TASK NUMBER C1330		
		5f. WORK UNIT NUMBER		
7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Institute for Defense Analyses 4850 Mark Center Drive Alexandria, Virginia 22311-1882		8. PERFORMING ORGANIZATION REPORT NUMBER NS D-14305 H 2020-000299/2		
9. SPONSORING / MONITORING AGENCY NAME(S) AND ADDRESS(ES) Institute for Defense Analyses 4850 Mark Center Drive Alexandria, Virginia 22311-1882		10. SPONSOR/MONITOR'S ACRONYM(S) IDA		
		11. SPONSOR/MONITOR'S REPORT NUMBER		
12. DISTRIBUTION / AVAILABILITY STATEMENT Approved for Public Release. Distribution Unlimited.				
13. SUPPLEMENTARY NOTES Project Leader, Ouellette, Stephen M.				
14. ABSTRACT This paper summarizes the orbital debris risk assessment process, and IDA research and support in various orbital debris risk assessment tasks performed for NASA and other organizations, from the Columbia disaster in 2003 through the present day.				
15. SUBJECT TERMS Orbital debris; spacecraft survivability				
16. SECURITY CLASSIFICATION OF:		17. LIMITATION OF ABSTRACT Unlimited	18. NUMBER OF PAGES 18	19a. NAME OF RESPONSIBLE PERSON Stephen Ouellette (SED)
a. REPORT Unclassified	b. ABSTRACT Unclassified	c. THIS PAGE Unclassified		19b. TELEPHONE NUMBER (include area code) (703) 845-2443